#### **TYRAP BASE**

DESC	CRIPTION OF THE ITEM	TYRAP BASE	
SPECIFICATION		TC-104	
PART NO. / DRAWING NO.		AE/TC-104	
SIZES		As per standard / requirement	
MATERIAL SPECIFICATION		NYLON	
OPERATING TEMPERATURE		$-55^{\circ}$ C to $+85^{\circ}$ C	
	SYSTEM	ELECTRICAL	
END USE	HELICOPTER / AIRCRAFT	Helicopters and other Aircraft wherever suitable	
CDL	LIFE	ON CONDITION	
TEST SCHEDULE / TEST ORDER NO.		As per Standard	
DEVELOPMENT AGENCY		M/s AEROSPACE ENGINEERS, SALEM	
PROVISIONAL CLEARANCE NO. & DATE		LOCAL TYPE CERTIFICATION COMMITTEE APPROVAL NO.	TYPE APPROVAL NO.
		RCMA(H/C)/LTCC/55-2/2006 Date: 21.12.2006	

## **TYRAP**

<b>DESCRIPTION OF THE ITEM</b>	TYRAP	
SPECIFICATION	AE -TY SERIES	
PART NO. / DRAWING NO.	AE – TY SERIES	0
SIZES	LENGTH 80MM TO 400MM	9
MATERIAL SPECIFICATION	NYLON 6	
OPERATING TEMPERATURE	- 55°C to +100°C	
SYSTEM	Loom Assy.	
HELICOPTER / AIRCRAFT	Helicopters and other Aircraft wherever suitable	
LIFE	ON CONDITION	
TEST SCHEDULE / TEST ORDER NO.	QTS/ALH/105 Date: 16.02.2009	
DEVELOPMENT AGENCY	M/s AEROSPACE ENGINEERS, SALEM	
PROVISIONAL CLEARANCE NO. & DATE	LOCAL TYPE CERTIFICATION COMMITTEE APPROVAL NO.	TYPE APPROVAL NO.
	RCMA(H/C)/LTCC/55-2/2009 Date: 04.06.2009	

<b>DESCRIPTION OF THE ITEM</b>		CLAMP WITH RUBBER	THE STATE OF THE S	
SPECIFICATION		MS 21919		
PART NO. / DRAWING NO.		AE/MS 21919		
SIZES		As per standard		
MATERIAL SPECIFICATION		AA2024/AU4GI/3.13 64 1.4544.9/ AISI304Z10C NT18-11 61D6, 31B6, HM4923, MIL-R-6855C Class-I Gr.60		
OPERATING TEMPERATURE		-45 deg to +135 deg		
ENID	SYSTEM	Structure / Integration of Systems and LRUs		
END USE	HELICOPTER / AIRCRAFT	Helicopters and other Aircraft wherever suitable		
CDL	LIFE	ON CONDITION		
	Γ SCHEDULE / Γ ORDER NO.	QTS/ALH/IND/026 Date: 16.09.2000		
DEVELOPMENT AGENCY		M/s AEROSPACE ENGINEERS, SALEM		
PROVISIONAL CLEARANCE NO. & DATE		LOCAL TYPE CERTIFICATION COMMITTEE APPROVAL NO.	TYPE APPROVAL NO.	
		RCMA(H/C)/LTCC/55-2/2006 Date: 21.12.2006	TA 1170, TA 1175, TA 1167, TA 1178 (for rubber compound)	

DESCRIPTION OF THE ITEM		CLAMP WITH RUBBER	
SPECIFICATION		LN 9490	
PART NO. / DRAWING NO.		AE/LN 9490	
SIZES		As per standard	
MATERIAL SPECIFICATION		1.4544.9/AISI304Z10C NT18-11 5.5688/61D6/MIL-R-25988B	
OPERATING TEMPERATURE		-45 deg +135 deg C	
END	SYSTEM	Structure/Integration of Systems & LRUs	
	HELICOPTER / AIRCRAFT	Helicopters and other Aircraft wherever suitable	
	LIFE	ON CONDITION	
TEST SCHEDULE / TEST ORDER NO.		QTS/ALH/IND/026 Date: 16.09.2000	
DEV	ELOPMENT AGENCY	M/s AEROSPACE ENGINEERS, SALEM	
PROVISIONAL CLEARANCE NO. & DATE		LOCAL TYPE CERTIFICATION COMMITTEE APPROVAL NO.	TYPE APPROVAL NO.
		RCMA(H/C)/LTCC/55-2/2006 Date: 21.12.2006	TA 1170, TA 1276 (for rubber compound)

<b>DESCRIPTION OF THE ITEM</b>		CLAMP WITH RUBBER	
SPECIFICATION		LN 9491	
PART NO. / DRAWING NO.		AE/LN 9491	
SIZES		As per standard	
MATERIAL SPECIFICATION		AA2024/AU4G1/3.1364 5.5688/61D6/MIL-R-25988B	
OPERATING TEMPERATURE		-45 deg +135 deg C	
		Structure / Integration Of Systems and LRUs	
END USE	HELICOPTER / AIRCRAFT	Helicopters and other Airborne Stores wherever suitable	
	LIFE	ON CONDITION	
TEST SCHEDULE / TEST ORDER NO.		QTS/ALH/IND/026 Date: 16.09.2000	
DEVELOPMENT AGENCY		M/s AEROSPACE ENGINEERS, SALEM	
PROVISIONAL CLEARANCE NO. & DATE		LOCAL TYPE CERTIFICATION COMMITTEE APPROVAL NO.	TYPE APPROVAL NO.
		RCMA(H/C)/LTCC/55-2/2006 Date: 21.12.2006	TA 1170, TA 1276 (for rubber compound)

<b>DESCRIPTION OF THE ITEM</b>		CLAMP WITH RUBBER	
SPECIFICATION		TL-212	
PART NO. / DRAWING NO.		AE/TL-212	
SIZES		As per standard	
MATERIAL SPECIFICATION		AISI 304, HM4923	
OPERATING TEMPERATURE		160°C	
	SYSTEM	POWER PLANT	
END USE	HELICOPTER / AIRCRAFT	Helicopters and other Aircraft wherever suitable	
	LIFE	ON CONDITION	
TEST SCHEDULE / TEST ORDER NO.		As per Standard	
DEVELOPMENT AGENCY		M/s AEROSPACE ENGINEERS, SALEM	
PROVISIONAL CLEARANCE NO. & DATE		LOCAL TYPE CERTIFICATION COMMITTEE APPROVAL NO.	TYPE APPROVAL NO.
		RCMA(H/C)/LTCC/55-2/2006 Date: 21.12.2006	TA 1167 (for rubber compound)

<b>DESCRIPTION O</b>	F THE ITEM	CLAMP WITH RUBBER	
SPECIFICATION		TLF-212	
PART NO. / DRAWING NO.		AE/TLF-212	
SIZES		As per standard	
MATERIAL SPECIFICATION		AISI 304, HM4923	
OPERATING TEMPERATURE		160°C	
SYSTEM		FLIGHT CONTROLS	
HELICOPT	TER / AIRCRAFT	Helicopters and other Aircraft wherever suitable	
LIFE		ON CONDITION	
TEST SCHEDULE / TEST ORDER NO.		As per Standard	
DEVELOPMENT AGENCY		M/s AEROSPACE ENGINEERS, SALEM	
PROVISIONAL CLEARANCE NO. & DATE		LOCAL TYPE CERTIFICATION COMMITTEE APPROVAL NO.	TYPE APPROVAL NO.
		RCMA(H/C)/LTCC/55-2/2006 Date: 21.12.2006	TA 1167 (for rubber compound)